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1 5 AUG 1962

MUCHANIUM	FOR:	Acting	Assistant	Director.	<b>CEA</b>

SUBJECT

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: Energency Reading Device

Mr. John Perengosky (opening remarks)

- 1. The operational requirement for an emergency heading device in the OKCART vehicle was throughly discussed and developed during a meeting 9 August 1962 at Project Headquarters. The purpose of this paper is to summarize the results of this investigation and to formalize the operational requirement.
  - 2. The participants of the meeting mentioned above included:

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	ting culminated several months' investigation into this	
ser amera	s which the following actions transpired:	
a.	Initial development coordination between and Lo	en terbie
<b>b.</b>	Commender requested assistance from Acting Dire	
		ULU
CHA WITH	regard to development of a suitable device.	
e.	was requested to conduct an informal	422-
	to the subject.	that it

- 3. The essential elements to recognize in this problem include the following:
  - a. The normal navigation heading devices include radio sids, the IRS and the MA-1 compass (magnetic or gyro mode). The possibility of system failure of each unit does exist, however the "design redundancy" should allow sufficient backup.
  - b. Lockheed established the point however, that the remote possibility exists that electrical failure is possible to the extent that all normal navigation references could be lost (alternator failure). Under this condition aircraft operation is still possible however heading reference must be obtained from an emergency source.

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- c. In the event of INS failure, the MA-1 compass is available for an emergency heading reference. If the failure occurs in areas other than Polar regions, the magnetic mode can be utilized directly. However, the unslaved or gyro mode will have to be utilized in Polar regions which presents several problems, one of which is the necessity of rather frequent heading checks to ascertain gyro drift and correct if necessary for Coriolis effects. In this instance an emergency heading device will be required.
- d. The backup Emergency Heading Device should include the following operational considerations:

Be independent of sircraft malfunctions.
 Be extremely simple to use by a pilot under emergancy

operational conditions.

(3) Furnish a heading reference under daylight and/or darkness conditions to within a tolerance of plus or minus 3 degroes.

(4) Product development abould be completed prior to 1

Jenuary, 1963.

	4.	The :	collowi	ng recom	bebre	solutio	n is b	ased on	the o	committe	re .
ward a	ar od	* the		faacqoen1	(which	discus	sed fir	re sepsi	rate i	solution	us),
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Both the and lockheed representatives felt that the device could be delivered in the required time and that the necessary aircraft installation problems could be satisfactorily resolved. The Staff Revigator concurred in the decision and indicated that this device will solve the problem. (Attachment #2 delineates the necessary tasks involved in the development of the device)

5. Your concurrence is requested to allow the Development Division to initiate actions necessary for the development, testing, and procurement of the "Celestial Saucer" emergency heading device.

SIGNED

Lt Colonel, USAF Chief, Operations Division, OSA 25X1

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JAMES A. CURNIMOHAM, JR. Acting Assistant Director (Special Activities)

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1. Corporation Proposal
2. Celestial Saucer Development Tasks

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OSA/GD/ rh (13Aug62)

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